

Seminari:
(by A. Farris)

RTBP III Libration Points Motion

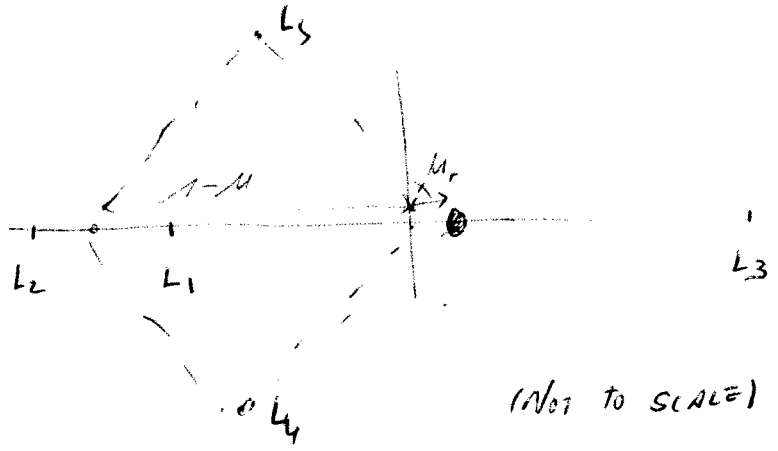
- Remember eq. motion:

$$\begin{aligned} \ddot{x} - 2\dot{y} &= \Omega_x, \\ \ddot{y} + 2\dot{x} &= \Omega_y, \\ \ddot{z} &= \Omega_z. \end{aligned}$$

on $\Omega(x, y, z) = \frac{1}{2}(x^2 + y^2) + \frac{(1-\mu)}{r_1} + \frac{\mu}{r_2} + \mu \frac{1-\mu}{2}$

- Eq. Points:

• there are five: L_i



in this talk we will focus on the motion around the collinear points. (L_1, L_2, L_3)

- L_1 and L_2 have been very studied and are very usefull for mission applications
- We want to understand the periodic and quasi-periodic motion around these points.
- As we saw in RTBP Part II these fixed points are of the type centre x centre x saddle.

Motion around the collinear eq points:

1st change of variables to bring the fixed point to the origin:

$$x = \bar{x} + \xi_j x + \mu + a_j$$

$$y = \bar{y} + \xi_j y$$

$$z = \xi_j z$$

or $\xi_j \equiv$ distance to the closest primary / solution of the Euler quartic

$$a_1 = -1 + \xi_1$$

$$a_2 = -1 - \xi_2$$

$$a_3 = \xi_3$$

(Note: Scaling is ~~also~~ done so that the expansion around the fixed point ~~is~~ has good numerical properties.)

Using Legendre Polynomials;

$$\frac{1}{\sqrt{(x-A)^2 + (y-B)^2 + (z-C)^2}} = \frac{1}{D} \sum_{n=0}^{\infty} \left(\frac{\rho}{D}\right)^n P_n \left(\frac{Ax + By + Cz}{D\rho}\right)$$

on $D^2 = A^2 + B^2 + C^2$, $\rho^2 = x^2 + y^2 + z^2$; P_n polynomial Legendre of degree n .

By expand:

$$\ddot{x} - 2\dot{y} - (1+2c_2)x = \frac{\partial}{\partial x} \sum_{n \geq 3} c_n(\rho) \rho^n P_n\left(\frac{x}{\rho}\right)$$

$$\ddot{y} + 2\dot{x} + (c_2+1)y = \frac{\partial}{\partial y} \sum_{n \geq 3} c_n(\rho) \rho^n P_n\left(\frac{x}{\rho}\right)$$

$$\ddot{z} + c_2 z = \frac{\partial}{\partial z} \sum_{n \geq 3} c_n(\rho) \rho^n P_n\left(\frac{x}{\rho}\right)$$

or

$$G_n(\mu) = \frac{1}{\sum_3^n} \left((+1)^n \mu + (-1)^n \frac{(1-\mu) \sum_3^{n+1}}{|1 + \sum_3^n|^{n+1}} \right) \text{ for } L_{1,2} \quad i, j = 1, 2$$

$$G_n(\mu) = \frac{(-1)^n}{\sum_3^n} \left(|1-\mu| + \mu \cdot \frac{\sum_3^{n+1}}{1 + \sum_3^n} \right) \text{ for } L_3$$

Stability of the fixed points (loop's & rep's)

$$J|_{\text{loop}} = \begin{bmatrix} 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \\ 1+c_2 & 0 & 0 & 0 & 2 & 0 \\ 0 & 1-c_2 & 0 & -2 & 0 & 0 \\ 0 & 0 & -c_2 & 0 & 0 & 0 \end{bmatrix}$$

⇒ Notice that the $\{z, \dot{z}\}$ plane motion is decoupled from the rest. $\{x, y, \dot{x}, \dot{y}\}$

$$\bullet \rho_0(\lambda) = \underbrace{[\lambda^4 + (2-c_2)\lambda^2 + (1+c_2-2c_2^2)]}_{\text{rest}} \cdot \underbrace{\lambda^2 + c_2}_{\text{associated to } z, \dot{z}}$$

taking $V = \lambda^2$ we can see that the roots are:

$$V_1 = \frac{c_2 - 2 - \sqrt{9c_2^2 - 8c_2}}{2}, \quad V_2 = \frac{c_2 - 2 + \sqrt{9c_2^2 - 8c_2}}{2}$$

As $c_2 > 1 \quad \forall \mu \in [0, \frac{1}{2}]$

$\Rightarrow V_1 > 0; V_2 < 0 \Rightarrow W_1 = \sqrt{-V_1}$ complex eigenvalues

$\lambda = \sqrt{V_2}$ real eigenvalues

$W_2 = \sqrt{-c_2}$ complex eigenvalues

\Rightarrow fixed points are ~~unstable~~
centre x centro x saddle.

One can easily compute the eigenvectors and see:

$$u_\lambda = (2\lambda, \lambda^2 - 2c_2 - 1, 0, 2\lambda^2, \lambda(\lambda^2 - (1 + 2c_2)), 0)$$

$$u_{-\lambda} = (-2\lambda, \lambda^2 - 2c_2 - 1, 0, 2\lambda^2, -\lambda(\lambda^2 - (1 + 2c_2)), 0)$$

$$u_{W_1} = (0, -W_1^2 - 1 - 2c_2, 0, -2W_1^2, 0, 0)$$

$$v_{W_1} = \frac{1}{W_1} (0, 0, 0, 0, -W_1(W_1^2 + 1 + 2c_2), 0)$$

$$u_{W_2} = (0, 0, 1, 0, 0, 0)$$

$$v_{W_2} = (0, 0, 0, 0, 0, 1)$$

- We have 2 central directions + 1 saddle plane.

Lyapunov Centre Theorem: Assume we have a system
 We have a non-degenerate n^{th} integral and an equilibrium point
 with $\pm \omega_1, \lambda_3, \dots, \lambda_n$ by $\text{Re} \lambda_j \neq 0 \in \mathbb{C}$ pure imaginary
 $\exists f_{\omega_j} \in \mathbb{R}^n \forall j=3, \dots, n \Rightarrow \exists$ one-parameter family
 of periodic orbits emanating from the eq. point

Assume that as $\omega_1, \omega_2 \notin \mathbb{Q} \Rightarrow$ we have two families of periodic orbits; one for each frequency ω_1 and ω_2

$\rightarrow \omega_1$ gives rise to the planar family (Lyapunov orbits)

$\rightarrow \omega_2$ gives rise to the vertical family (Vertical Lyap. orbits)

We can use a continuation method to follow these families

(\rightarrow note: very unstable: suitable to use Poincaré Shooting; other option Lindstedt Poincaré).

• LINEAR APPROXIMATION OF MOTION AROUND EQUILIBRIA:

$$\begin{aligned}
 X(t) = & A e^{\lambda t} \cdot \vec{u}_\lambda + B e^{-\lambda t} \cdot \vec{u}_{-\lambda} + \\
 & + (C \cos(\omega_1 t + \phi_0) \vec{u}_{\omega_1} + \sin(\omega_1 t + \phi_0) \vec{v}_{\omega_1}) + \\
 & + D (\cos(\omega_2 t + \phi_0) \vec{u}_{\omega_2} + \sin(\omega_2 t + \phi_0) \vec{v}_{\omega_2})
 \end{aligned}$$

on $A, B, C, D, \phi_0, \phi_0$ given by the initial condition at $t=t_0$.

→ Taking: $A = B = C = 0 \rightarrow$ Vertical family

$A = B = D = 0 \rightarrow$ Planar family.

• STABILITY AROUND PERIODIC ORBIT:

- T-Periodic orbit $\Rightarrow \Phi_t(x) \equiv$ flow around trajectory
 and $M = D\Phi_T(x_0) \equiv$ monodromy matrix
 for the periodic orbit x_0 .

As the system is Hamiltonian there is 1st Integral

$$\Rightarrow \text{Spect } M = \{ \lambda_1, \lambda_1, \lambda_1^{-1}, \lambda_1^{-1}, \lambda_2, \lambda_2^{-1} \}$$

• We define the stability parameters $S_j = \lambda_j + \lambda_j^{-1}$ $j=1, 2$.

→ Hyperbolic: if $S_j \in \mathbb{R}, |S_j| > 2$ ($\lambda_j \in \mathbb{R} \setminus \{1, -1\}$)

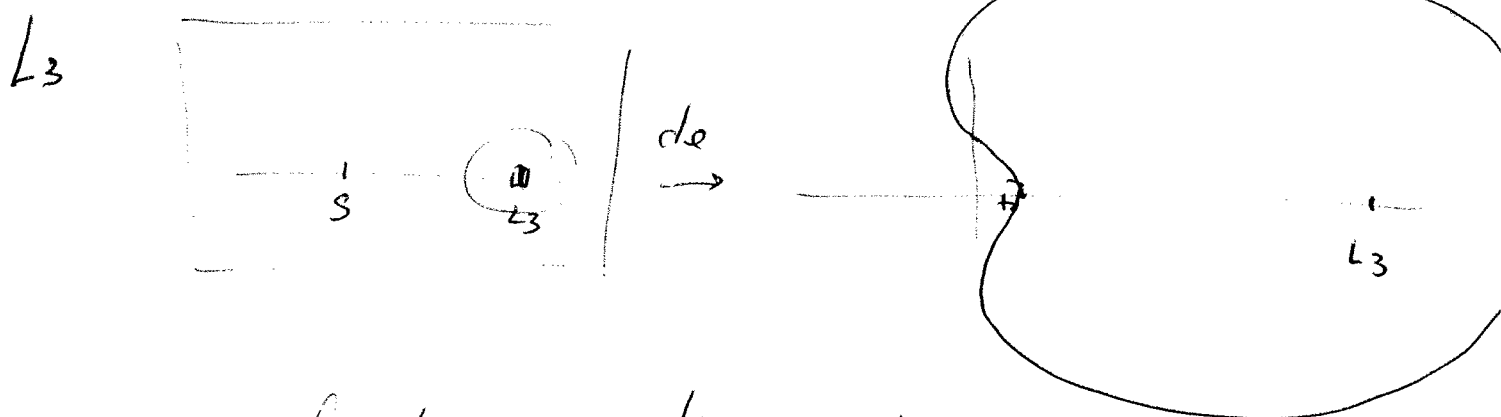
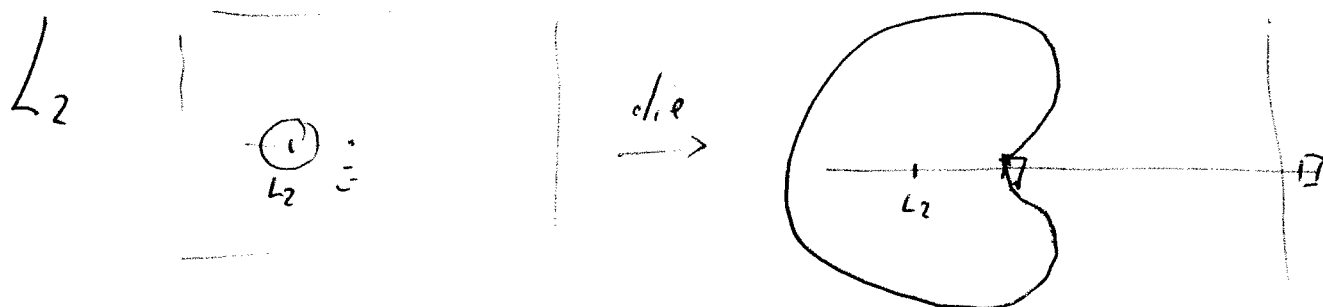
Elliptic: if $S_j \in \mathbb{C}, |S_j| < 2$ ($\lambda_j = e^{i\theta}$)

~~more~~
- We have followed the Poincaré and Vertical Lyapunov
families for L_1 and L_2 , for $\mu =$
(μ for Sun-Earth model)

- Special attention to 1st bifurcations + where they die.

Classical Periodic Orbits.

PLANAR FAMILIES:



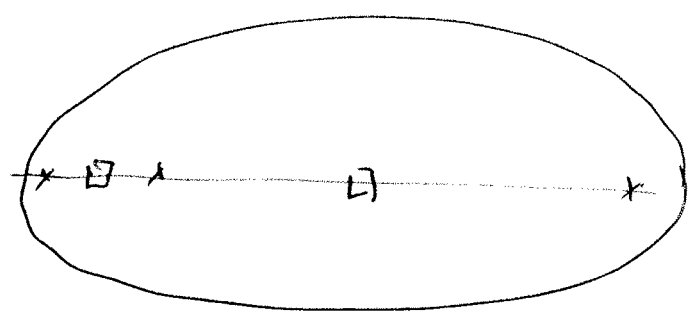
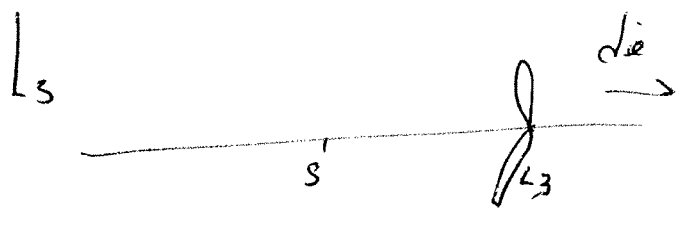
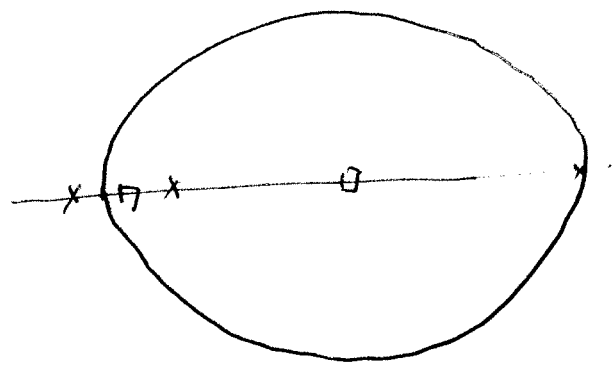
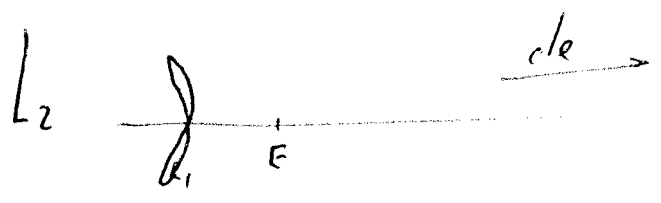
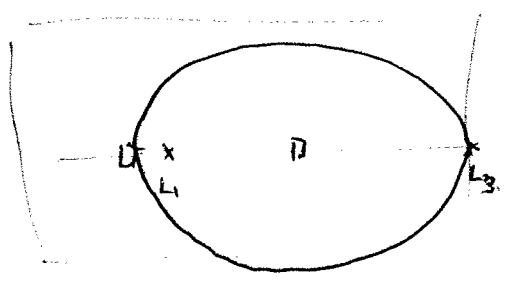
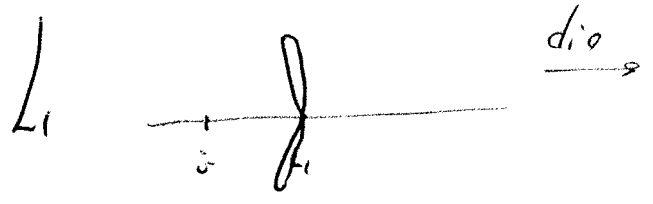
- L_i planar family dies with a collision with one of the two primaries.

- At the beginning L_i planar family bifurcates.

$C \times S \rightarrow S \times S$ and the Halo orbits REPEAR.

(Show/See present OP_librc.pdf)

VERTICAL FAMILIES:



L_1 formants at: orbit surrounding massive body & L_1 L_3

L_2 " " : orbit surrounding two bodies & L_1 L_3

L_3 " " " " two bodies & L_1 L_2 L_3

• Δ bifurcation Apoc at the beginning

(See prot OP-libcc.pdf)

Heliciformes:

- They appear ~~when~~ when PLENNAR ORBIT bifurcates.

- Symmetric w.r.t $y=0$ & $z=0$.

- With this we can see the families of invariant tori that appear around the eq. point

$$H = \frac{1}{2} \dot{x}_1^2 + \frac{1}{2} \dot{y}_1^2 + i\omega_1 \frac{(x_2^2 + y_2^2)}{2} + i\omega_2 \frac{(x_3^2 + y_3^2)}{2}$$

as $\omega_1, \omega_2 > 0 \Rightarrow$ for a fixed $H=h$ energy level the motion is bounded by an ellipsoid on the centre manifold (taking $x_i = y_i = 0$)

- look at pictures of the centre manifold for different h .

\rightarrow As h increases \Rightarrow planar loops bifurcate and H tori orbits appear.

\rightarrow We see all that can be here.

[Plots]

(two dif sections $z=0, \dot{z}(h)$; $x_3=0, \dot{x}_3(h)$; $y=0, \dot{y}(h)$ at 1st order ; $x_1=0, \dot{x}_1(h)$)

• To have a more complete understanding of the motion it is interesting to do the so-called REDUCTION TO THE CENTRE MANIFOLD.

• Due to the big instability of the eq. points, Poincaré sections & numerical integration gives problems as the trajectories quickly escape from a close vicinity of the eq. point.

• By reducing to the centre manifold:

(1) we decouple the elliptic directions from the hyperbolic ones up to high order.

(2) use the high order approximation of the centre manifold to do numerical integrations

(3) How to do this: (SEE REF)